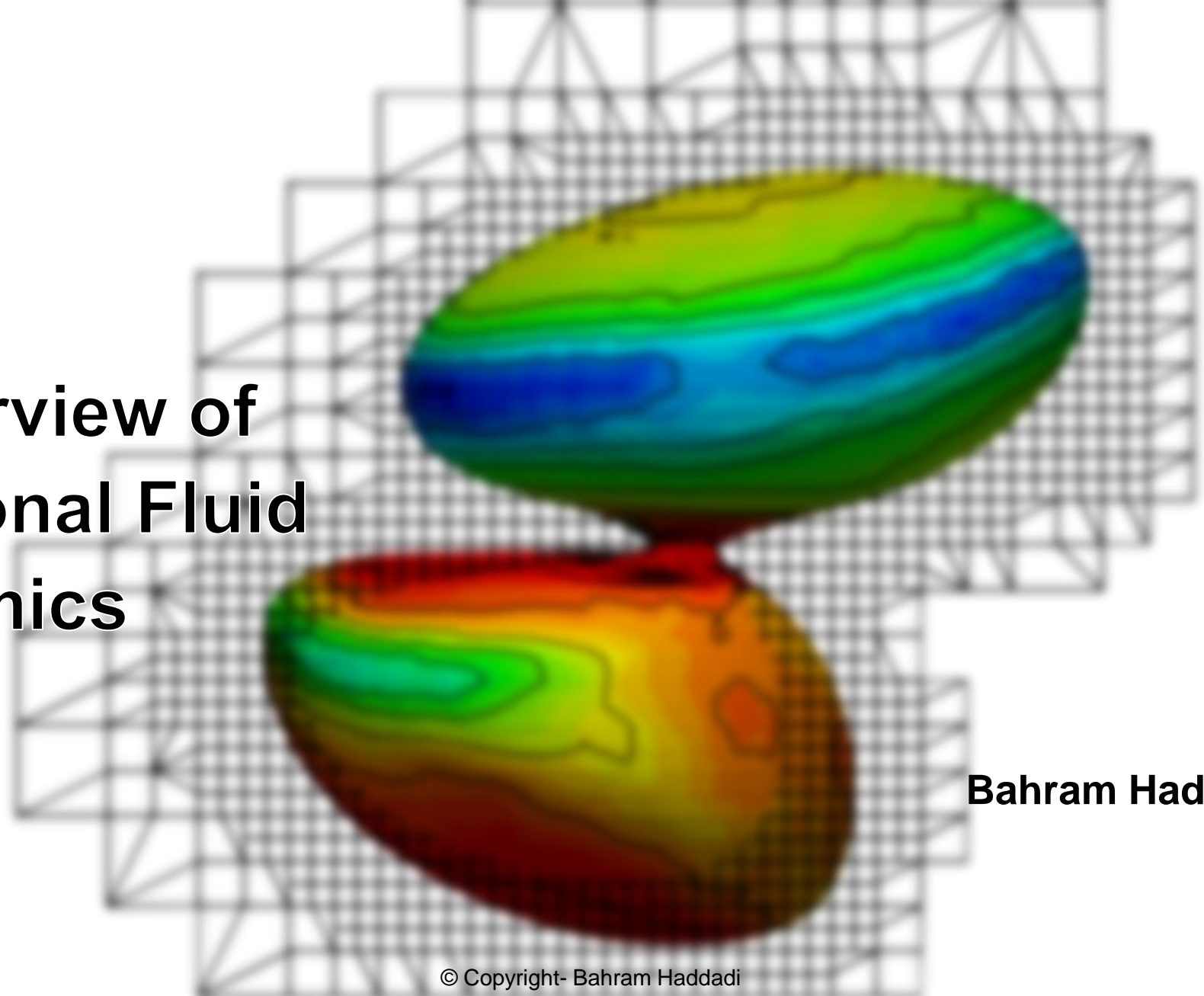


Brief Overview of Computational Fluid Dynamics

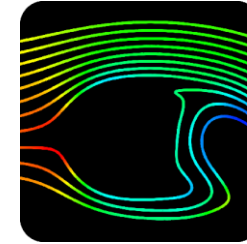


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What is Computational Fluid Dynamics

- Behavior of fluidic systems
 - Experimental investigations
 - Numerical approaches
 - Analytical solutions
 - Numerical representation → CFD
 - Combination of Physics and Numeric
- Computer based analysis of systems
 - fluid flow
 - Heat transfer
 - Mass transfer, ...
- Application range
 - Industrial
 - non-industrial

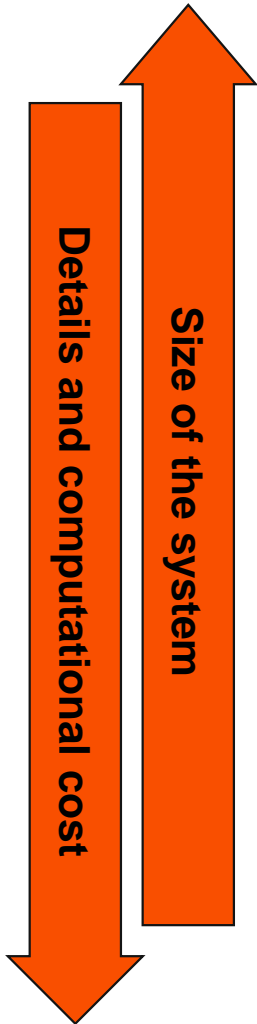


**COMPUTATIONAL
FLUID
DYNAMICS**

- **Not a tool, but a science**
- The technique is very powerful
 - If combined with fluid dynamics and numerics knowledge



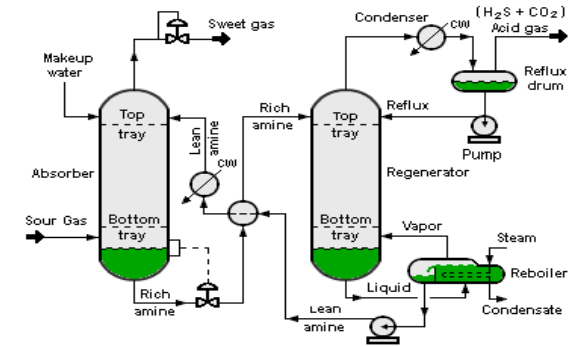
CFD vs. Other Numerical Methods



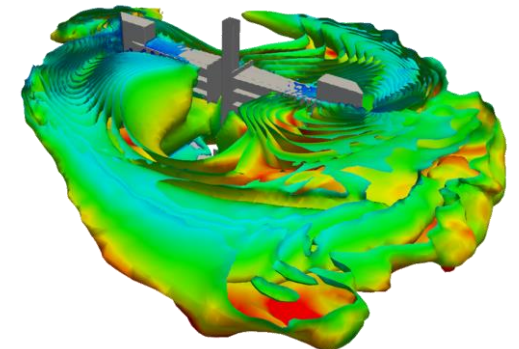
- Black box modeling
 - Just time resolved
 - E.g. balance calculations in a system
 - Very fast but least information provided
- Process simulation
 - Time resolved
 - One space dimension also resolved
 - Reasonably fast and moderate details about the system
 - Suitable for complex industrial plants and pilots
- Computational fluid dynamics
 - Time resolved
 - Fully space resolved
 - Suitable for detailed systems analysis



<https://en.wikipedia.org>



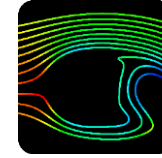
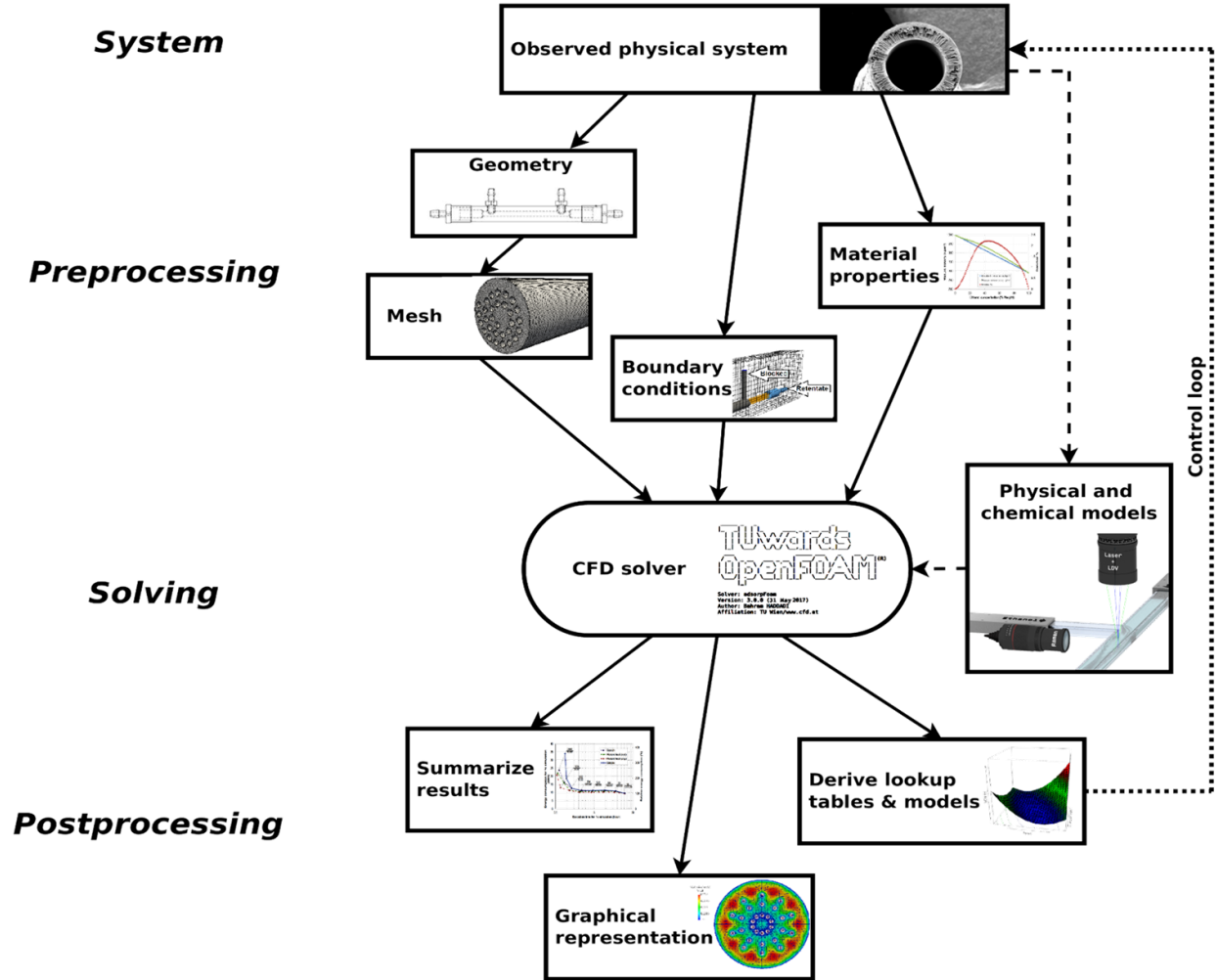
<https://en.wikipedia.org>



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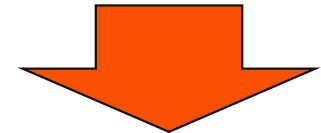
CFD Workflow



COMPUTATIONAL



EXPERIMENTAL



ENGINEERING
FLUID
DYNAMICS

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CFD Tools

- Common CFD code categories
 - Commercial
 - Annual license cost
 - No/limited access to the code
 - Mostly user friendly interface and official support
 - E.g. Ansys Fluent, Ansys CFX, Star CCM+, **COMSOL**, ...
 - Free
 - No license cost
 - Mainly open-source
 - Missing official support or paid support
 - E.g. OpenFOAM®
- **All are built upon the same concept**





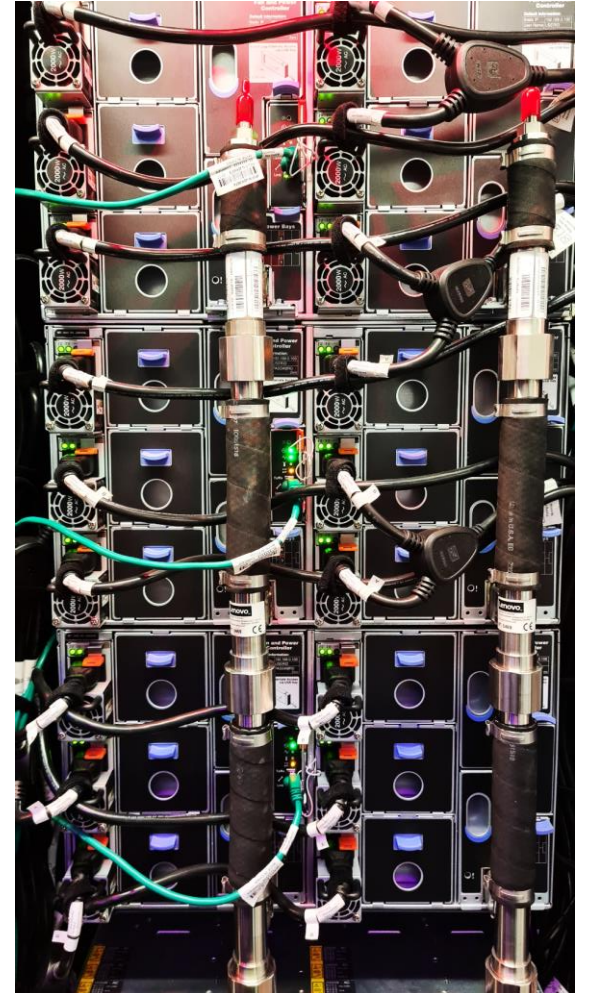
- Open Field Operation And Manipulation
 - Original development: 1980s – Imperial College (UK)
 - Open source CFD toolbox – written in C++
 - Operating system:
 - **Linux** – original development and best compatibility
 - **Windows** – using Docker
 - **OSX** – direct compilation or using Docker
 - Different flavours:
 - **ESI version**: openfoam.com – v2212
 - **Foundation version**: openfoam.org – openfoam10
 - **OpenFOAM extend**: Hevoje Jasak – foam-extend-4.1
 - Released for free download (source and binary)
 - Under General Public License





Why OpenFOAM®

- Advantages over commercial codes:
 - Open source and free → No license fee
 - Highly parallel
 - Users can inspect, alter, expand the source code
 - Users have to understand and know what he/she is doing, not just “clicking”!!!
- Advantages over open source codes:
 - One of the most versatile open source CFD programs
 - All standard finite volume algorithms implemented.
 - Industrial interest
 - Large community on the internet: cfd-online.com and stackoverflow.com



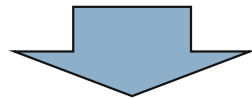
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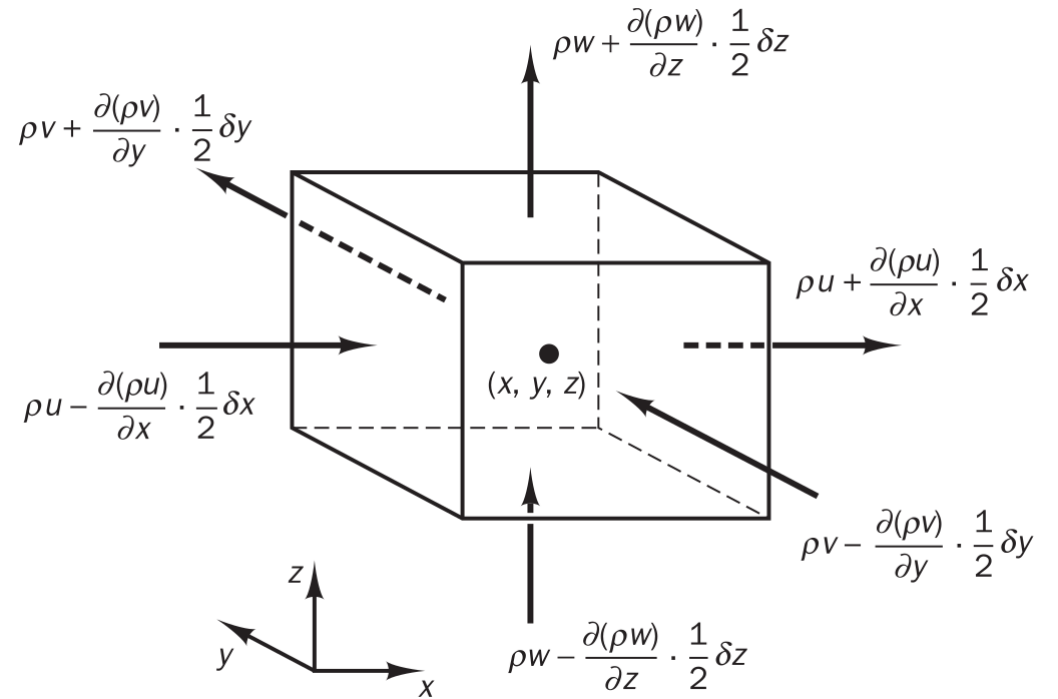
Mass Conservation – Continuity Equation

Accumulation = Input - Output

$$\begin{aligned} & \left(\rho u - \frac{\partial(\rho u)}{\partial x} \frac{1}{2} \delta x \right) \delta y \delta z - \left(\rho u + \frac{\partial(\rho u)}{\partial x} \frac{1}{2} \delta x \right) \delta y \delta z \\ & + \left(\rho v - \frac{\partial(\rho v)}{\partial y} \frac{1}{2} \delta y \right) \delta x \delta z - \left(\rho v + \frac{\partial(\rho v)}{\partial y} \frac{1}{2} \delta y \right) \delta x \delta z \\ & + \left(\rho w - \frac{\partial(\rho w)}{\partial z} \frac{1}{2} \delta z \right) \delta x \delta y - \left(\rho w + \frac{\partial(\rho w)}{\partial z} \frac{1}{2} \delta z \right) \delta x \delta y \\ & = \frac{\partial \rho}{\partial t} \delta x \delta y \delta z \end{aligned}$$



$$\frac{\partial \rho}{\partial t} + \frac{\partial(\rho u)}{\partial x} + \frac{\partial(\rho v)}{\partial y} + \frac{\partial(\rho w)}{\partial z} = 0 \quad \longrightarrow \quad \frac{\partial \rho}{\partial t} + \text{div}(\rho \mathbf{u}) = 0$$



An introduction to computational fluid dynamics – The finite volume method – 2nd Edition, Versteeg and Malalasekera 2007



Governing Equations

Continuity $\frac{\partial \rho}{\partial t} + \text{div}(\rho \mathbf{u}) = 0$

x -momentum $\frac{\partial(\rho u)}{\partial t} + \text{div}(\rho u \mathbf{u}) = -\frac{\partial p}{\partial x} + \text{div}(\mu \text{ grad } u) + S_{Mx}$

y -momentum $\frac{\partial(\rho v)}{\partial t} + \text{div}(\rho v \mathbf{u}) = -\frac{\partial p}{\partial y} + \text{div}(\mu \text{ grad } v) + S_{My}$

z -momentum $\frac{\partial(\rho w)}{\partial t} + \text{div}(\rho w \mathbf{u}) = -\frac{\partial p}{\partial z} + \text{div}(\mu \text{ grad } w) + S_{Mz}$

Energy $\frac{\partial(\rho i)}{\partial t} + \text{div}(\rho i \mathbf{u}) = -p \text{ div } \mathbf{u} + \text{div}(k \text{ grad } T) + \Phi + S_i$

Equations of state $p = p(\rho, T)$ and $i = i(\rho, T)$

e.g. perfect gas $p = \rho R T$ and $i = C_v T$

$$\frac{\partial(\rho \varphi)}{\partial t}$$

Time derivative

$$\nabla \cdot (\rho \varphi \mathbf{u})$$

Convection term

$$\nabla \cdot (\Gamma \nabla \varphi)$$

Diffusion term

$$S_\varphi$$

Source term

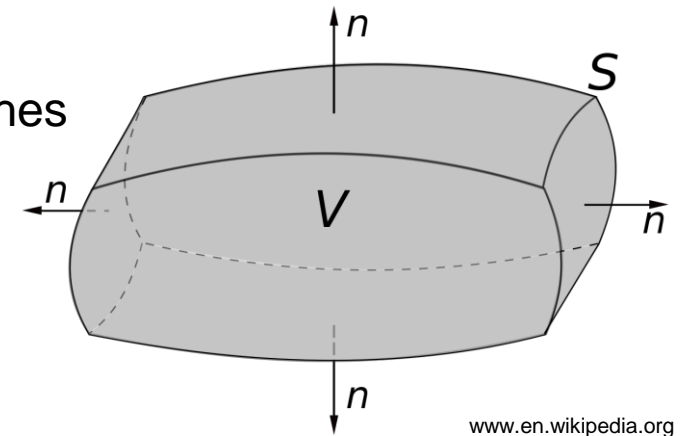


Finite Volume Approach

- **Mesh (grid):** Converting the domain into discrete domains
- **Grid cell:** The small volume surrounds each node of the mesh
- **Key step:** integration of the transport equation over a three-dimensional control volume
- **Gauss divergence theorem:** Replacing volume integral of the divergence term by surface integral
 - Terms evaluated as fluxes at the surfaces
 - Ensures the conservation of fluxes entering and exiting the grid
 - Allows for easy formulation of the balances on unstructured meshes
- Time-dependency: integration with respect to time

$$\frac{\partial(\rho\varphi)}{\partial t} + \nabla \cdot (\rho\varphi\mathbf{u}) = \nabla \cdot (\Gamma\nabla\varphi) + S_\varphi$$

$$\int_{\Delta t} \frac{\partial}{\partial t} \left(\int_{CV} \rho\varphi dV \right) dt + \int_{\Delta t} \int_A \mathbf{n} \cdot (\rho\varphi\mathbf{u}) dA dt = \int_{\Delta t} \int_A \mathbf{n} \cdot (\Gamma\nabla\varphi) dA dt + \int_{\Delta t} \int_{CV} S_\varphi dV dt$$



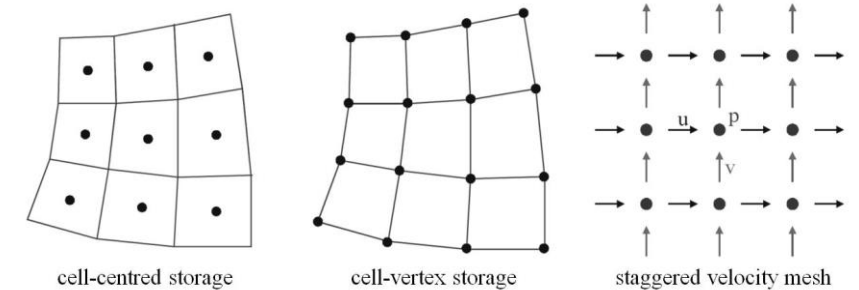
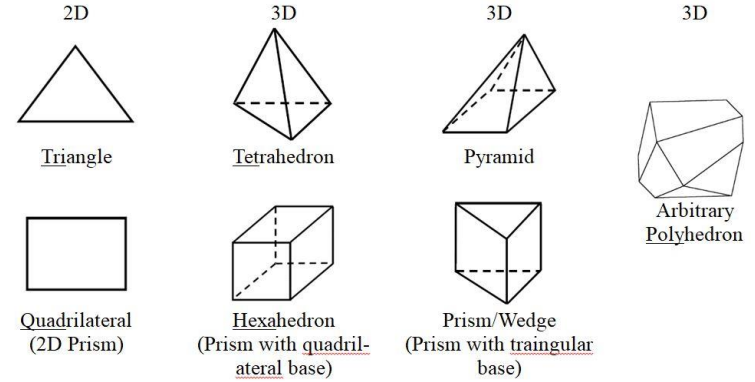
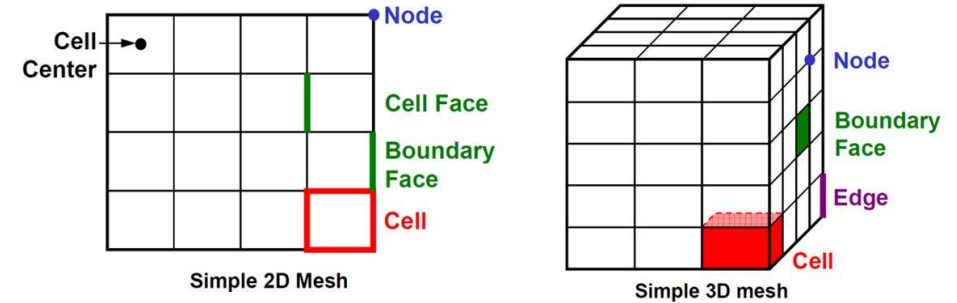
www.en.wikipedia.org

An introduction to computational fluid dynamics – The finite volume method – 2nd Edition, Versteeg and Malalasekera 2007



Mesh Terminology

- Cell: control volume created by domain discretization
- Node: grid point
- Cell center: center of a cell
- Edge: boundary of a face
- Face: boundary of a cell
 - Internal
 - Boundary
- Zone: grouping of nodes, faces, cells
- Domain: group of node, face and cell zones
- Nodes positions relative to the vertices
 - Cell centered
 - Cell vertex
 - Staggered

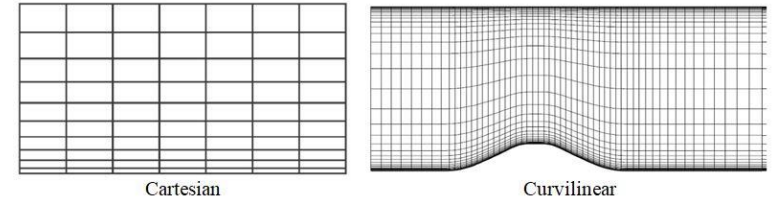


www.manchestercfd.co.uk/post/all-there-is-to-know-about-different-mesh-types-in-cfd



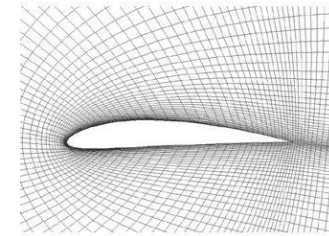
Mesh Types

- **Structured Grids**
 - Cartesian
 - Grid lines are always parallel to the coordinate axes
 - Curvilinear
 - Coordinate surfaces are curved to fit boundaries
 - *Orthogonality: all grid lines cross at 90°*
- **Block-structured Grids**
 - Matching
 - The grid on the common boundaries between the regions is the same
 - Non-matching
 - Chimera (overset)

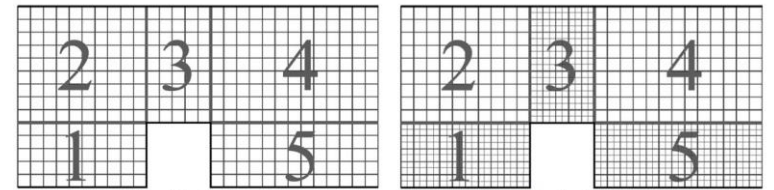


Cartesian

Curvilinear

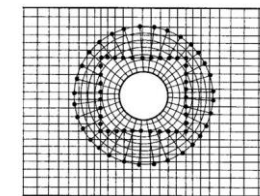


Curvilinear (Body-fitted)

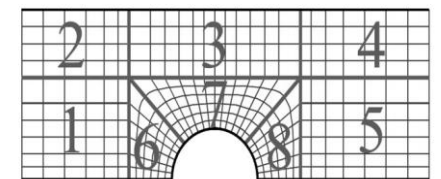


Matching block-structures

Non-matching block-structures (2 to 1 here)



Chimera



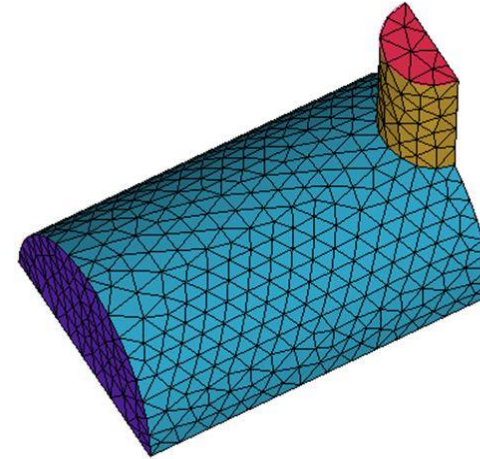
Non-orthogonal block-structured

www.manchestercfd.co.uk/post/all-there-is-to-know-about-different-mesh-types-in-cfd

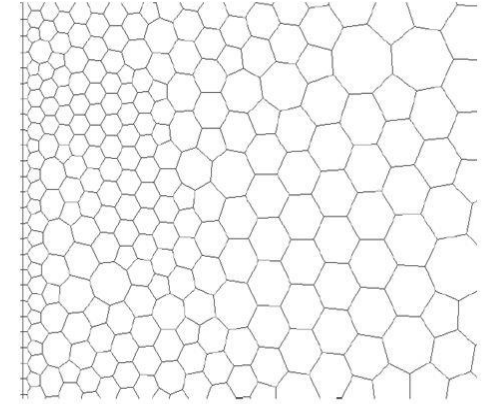


Mesh Types

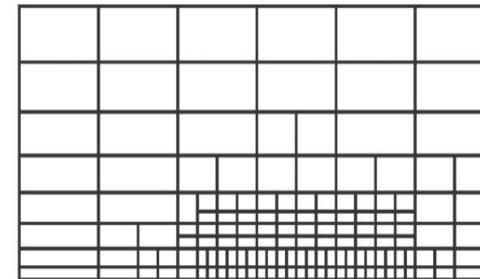
- **Unstructured Grids**
 - Triangular (tetrahedral)
 - Robust meshing algorithms
 - Quadrilateral (hexahedral)
 - Best for CFD calculations
 - Polygon (polyhedral)
 - Similar to tet, with less computational overhead
 - Hybrid
- **Quad/Hex advantages vs Tri/Tet**
 - Higher quality solutions for flow-aligned problems
 - Fewer cells/nodes than a comparable tri/tet mesh.
 - Reduced numerical diffusion



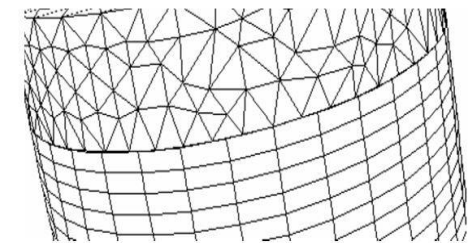
Triangular/tetrahedral



Polygon (polyhedral)



Quadrilateral (hexahedral)



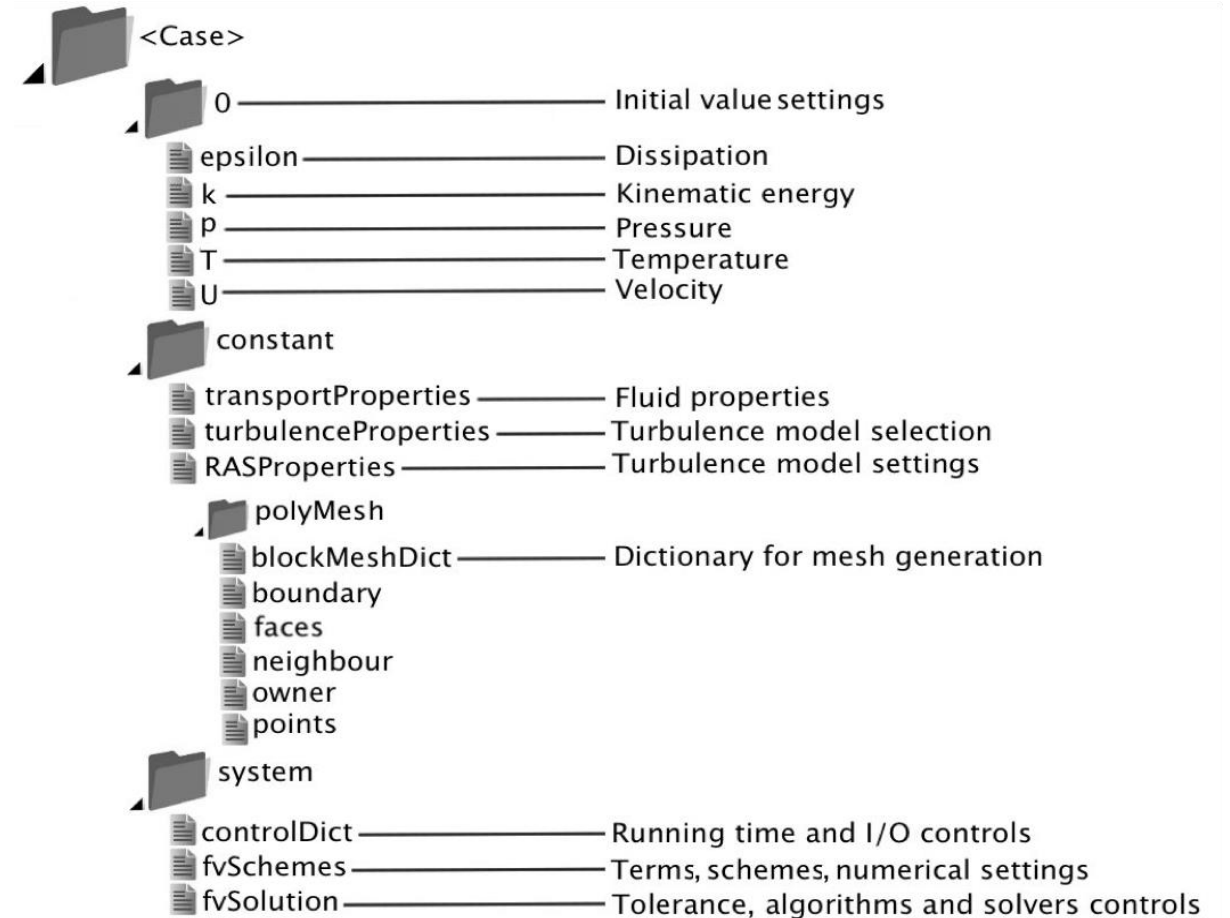
Hybrid Tetrahedral

www.manchestercfd.co.uk/post/all-there-is-to-know-about-different-mesh-types-in-cfd



OpenFOAM® Applications and Case Structure

- Solvers
 - Solving a specific continuum mechanics problem
 - icoFoam
 - simpleFoam
 - ...
- Utilities
 - Performing pre and post processing
 - Mesh preparation: blockMesh, snappyHexMesh
 - Simulation set-up: topoSet, setFields
 - Processing the results: postProcess
 - ...

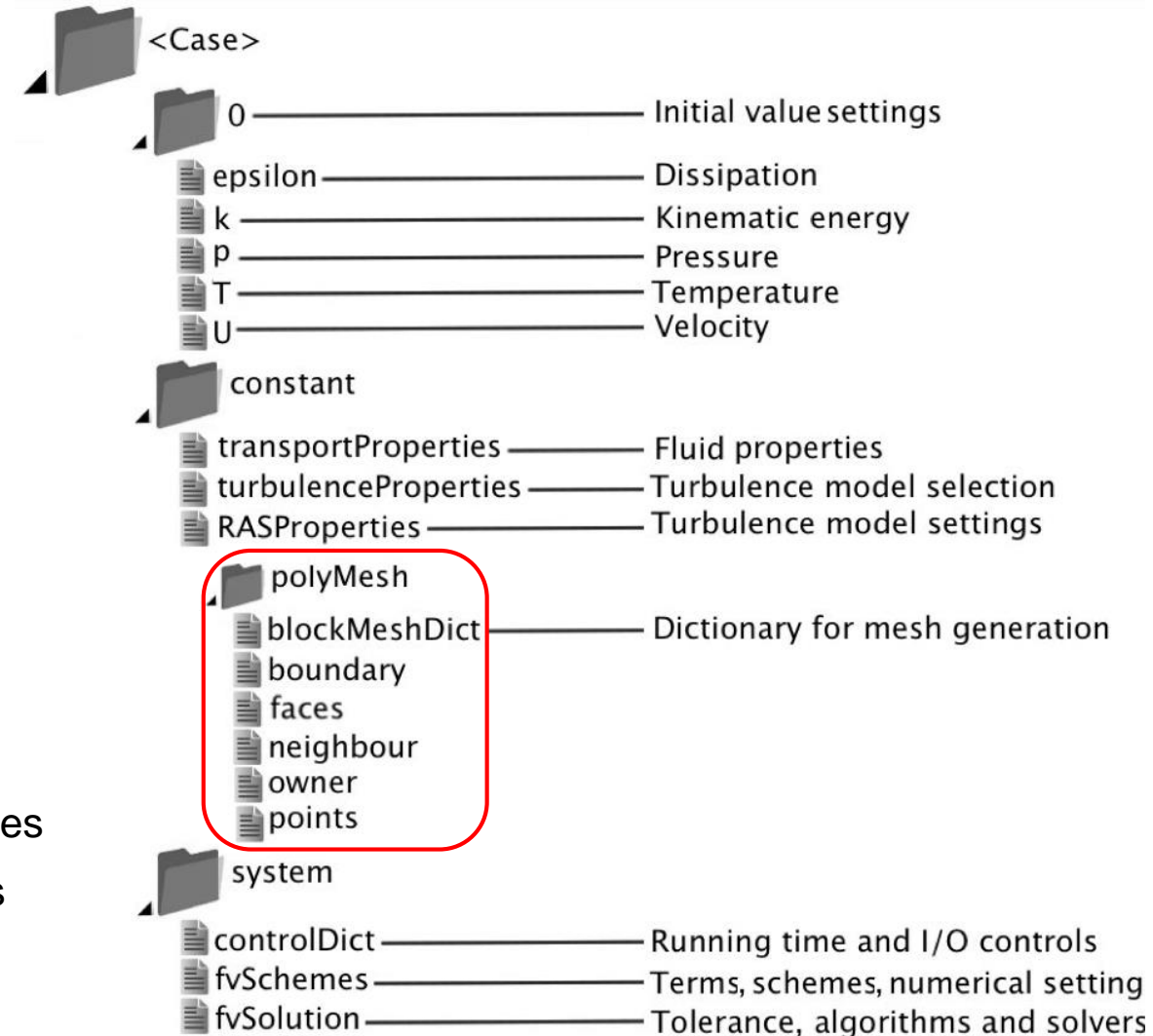


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Mesh in OpenFOAM[®]

- Unstructured mesh design
 - Capable of handling structured
- Basic mesh generation
 - *blockMesh*
 - Block structured meshes
 - Curved internal and external boundaries.
 - The mesh is setup in a script (no GUI)
 - Good for simple geometries
- Advanced mesh generation
 - *snappyHexMesh, cfMesh*
 - Automatic mesh generator for complex geometries
 - Possibility of mesh refinement at desired regions
 - Parallel
- Mesh import
 - *gambitToFoam, ansysToFoam, cfxToFoam, ...*



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Initial Conditions

- Starting value for the solver and once specified
 - Transient simulations
 - Initial state of the system
 - Next time step values are calculated based on these
 - Steady state simulations
 - Initial guess for the numerical system
 - Initial values will be replaced by newly calculated values
 - Better the initial values → the faster the convergence
- Value is assigned to the center of every cell
 - A uniform value for the whole field
 - Individual value per cell
 - Patching fields
 - OpenFOAM®: *setFields*
 - *Multi-solver solutions*

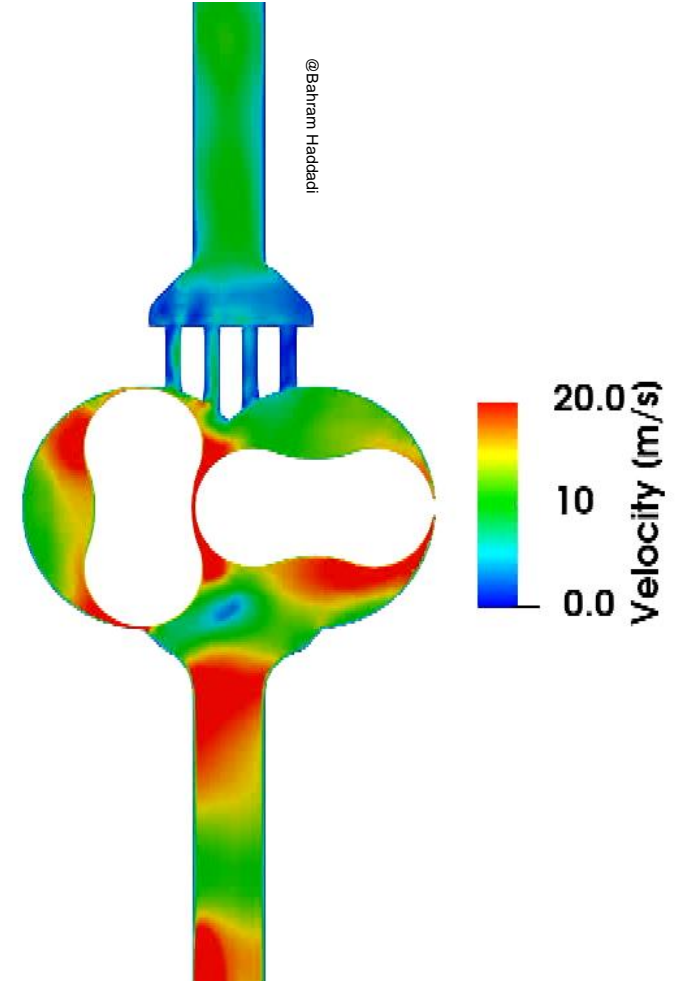


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Boundary Conditions

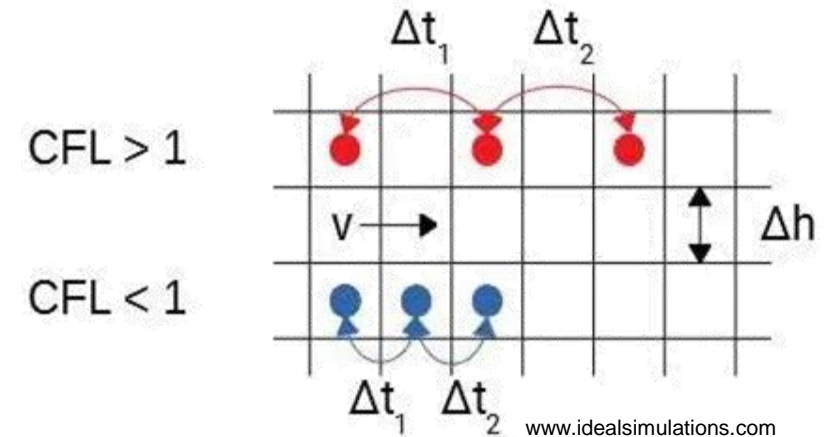
- Boundary conditions will connect the simulation domain with its surroundings
 - Each property interaction with outside the domain
- The values specified are located at the boundary faces of the domain
- Three main types of boundary conditions
 - **Dirichlet**: fixed value on the boundary
 - **Neumann**: fixed gradient on the boundary
 - **Mixed**: combined fixed value and gradient
- Most boundary conditions are either steady state or transient
- Ill-defined boundary conditions
 - Non-convergence
 - Incorrect results





Time Stepping and CFL condition

- Time step
 - How much the information travels across a computational grid cell
 - Too big time step
 - Information propagates through more than one grid cell
 - Numerically: Inaccurate solution - divergence
 - Physically: Nonphysical results
 - Too small time step
 - Computationally expensive
- Information transport should not “overtake” the physical transport
 - Courant-Friedrichs-Lewy (CFL) condition – Co
 - One-dimensional: $Co = u \frac{\Delta t}{\Delta x}$
 - Less or equal to one
 - Finer mesh \rightarrow smaller time step
 - Initial time step: maximum velocity and smallest mesh size





Discretization

- General transport equation

$$\frac{\partial(\rho\varphi)}{\partial t} + \nabla \cdot (\rho\varphi\mathbf{u}) = \nabla \cdot (\Gamma\nabla\varphi) + S_\varphi$$

- Time derivative

- Integrating over volume
- Euler implicit scheme

$$\int_V \frac{\partial\rho\varphi}{\partial t} dV \approx \frac{\rho_P^n \varphi_P^n - \rho_P^0 \varphi_P^0}{\Delta t} V_P$$

- Convection term

- Integrating over volume
- Applying Gauss's theorem
- φ_f to be calculated \rightarrow schemes

$$\int_A \mathbf{n} \cdot (\rho\varphi\mathbf{u}) dA \approx \sum_f \mathbf{n} \cdot (A\rho\mathbf{u})_f \varphi_f = \sum_f F \varphi_f$$

- Diffusion term

- Similar to convection term
- $\nabla_f\varphi$ gradient at the face
- Second order accurate \rightarrow \mathbf{d} orthogonal between P and N

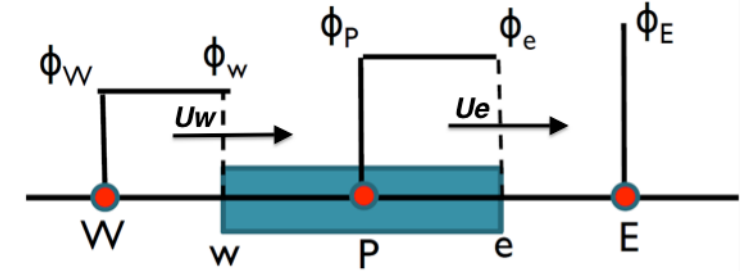
$$\int_A \mathbf{n} \cdot (\Gamma\nabla\varphi) dA = \sum_f \Gamma_f (\mathbf{n} \cdot \nabla_f\varphi) A_f$$
$$\mathbf{n} \cdot \nabla_f\varphi = \frac{\varphi_N - \varphi_P}{|\mathbf{d}|}$$



Common Discretization Schemes

- **First Order Upwind**

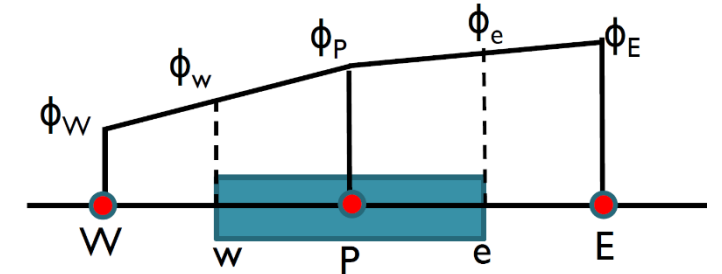
$$\begin{aligned} \varphi_e &= \varphi_P & \text{if, } F_e > 0 \\ \varphi_e &= \varphi_E & \text{if, } F_e < 0 \end{aligned}$$



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- **Central Differencing Scheme**

$$\varphi_e = \frac{\varphi_E + \varphi_P}{2}, \quad \varphi_w = \frac{\varphi_P + \varphi_W}{2}$$

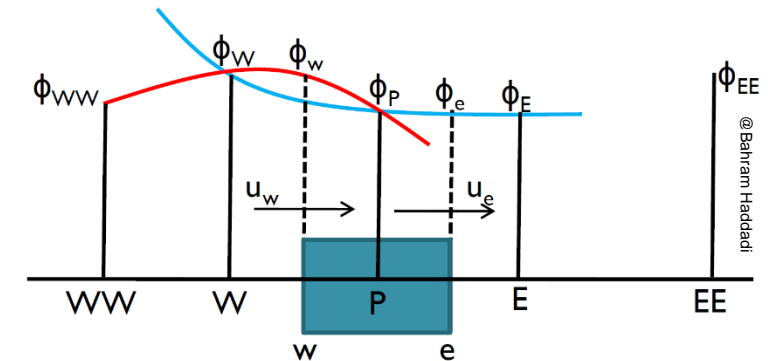


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- **Quadratic Upwind Interpolation for Convective Kinetics (QUICK)**

$$\text{When } F_e > 0, \quad \varphi_e = \frac{6}{8} \varphi_P + \frac{3}{8} \varphi_E - \frac{1}{8} \varphi_W$$

$$\text{When } F_w > 0, \quad \varphi_w = \frac{6}{8} \varphi_W + \frac{3}{8} \varphi_P - \frac{1}{8} \varphi_{WW}$$



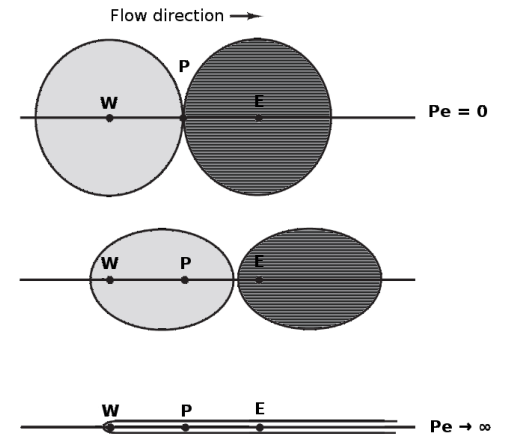
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Discretization Schemes Properties

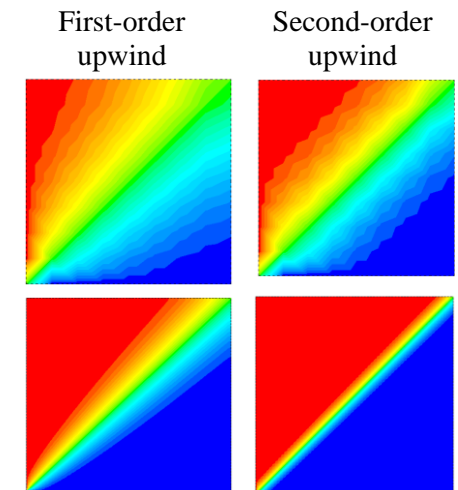
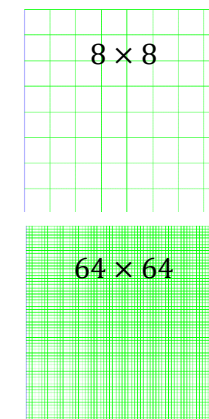
Prerequisite for results to be physically realistic

- **Conservativeness:** flux across a certain face must be equal the adjacent control volume flux through the same face
- **Boundedness:** ensures the solution remains within certain bound limits and guarantees the solution does not exhibit unphysical or unrealistic behavior
- **Transportiveness:** Peclet number, Pe . It measures the relative strengths of convection, N_{conv} and diffusion, N_{diff} .



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False diffusion: a multidimensional phenomenon and it occurs when the flow is not perpendicular to the grid lines. It is a numerically introduced diffusion and arises in convection dominated flows



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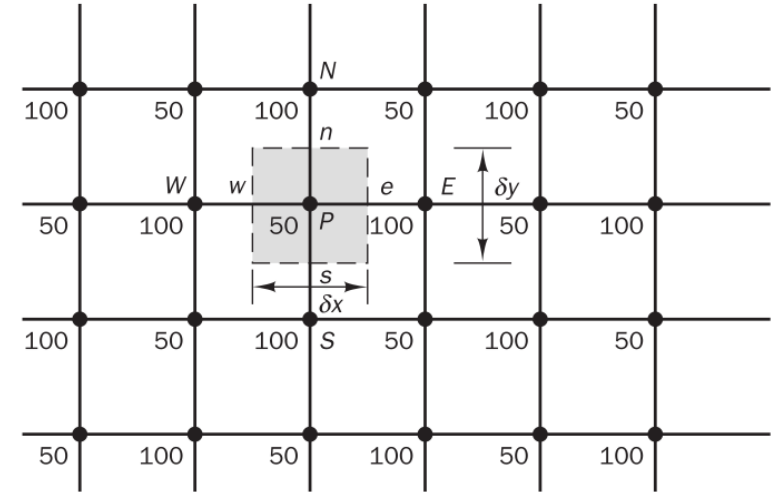
Discretization Schemes Assessment

Scheme	Conser- -vative	Bounded	Accuracy	Trans- -portive	Remarks
Upwind	Yes	Unconditionally bounded	First order	Yes	Include false diffusion if the velocity vector is not parallel to one of the coordinate directions
Central Differencing	Yes	Conditionally bounded*	Second order	No	Unrealistic solutions at large Pe number
QUICK	Yes	Unconditionally bounded	Third order	Yes	Less computationally stable. Can give small undershoots and overshoots



Variables Storage

- Variables evaluation
 - Checker-board problem
 - The cell pressure is not considered in the pressure gradient
 - Using Staggered grid
 - Scalar
 - Pressure, temperature, density etc.
 - At ordinary nodal points
 - Vectors
 - Velocity
 - At **Staggered** grid, centered around cell faces
 - Generation of velocities at exactly the locations where they are needed

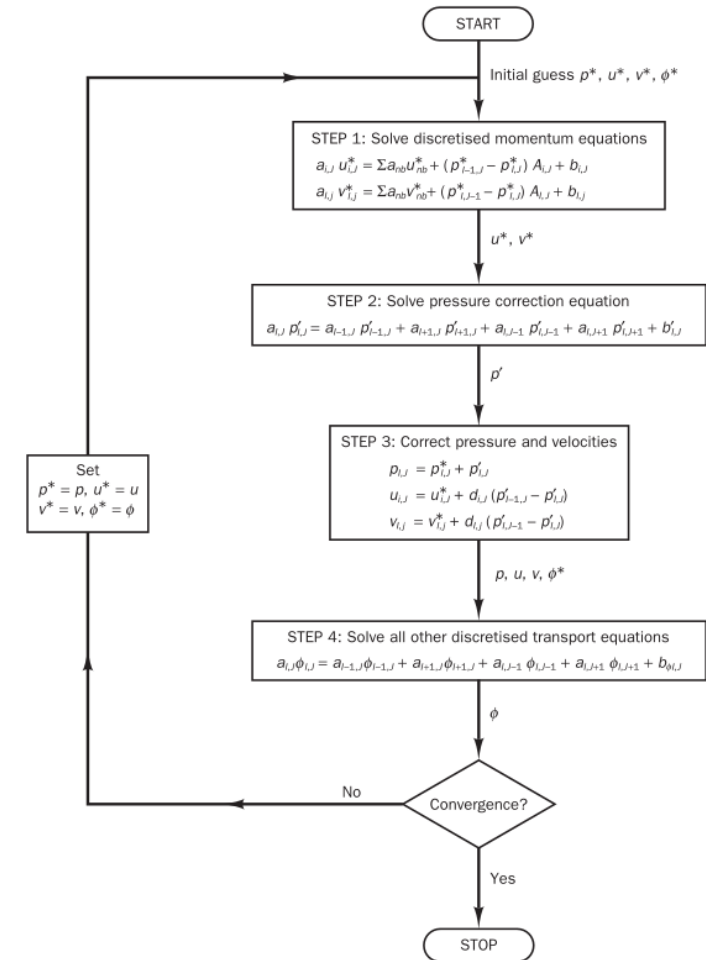


$$\frac{\partial p}{\partial x} = \frac{p_e - p_w}{\delta x} = \frac{\left(\frac{p_E + p_P}{2}\right) - \left(\frac{p_P + p_W}{2}\right)}{\delta x} = \frac{p_E - p_W}{2\delta x}$$



Pressure – Velocity Coupling Algorithms

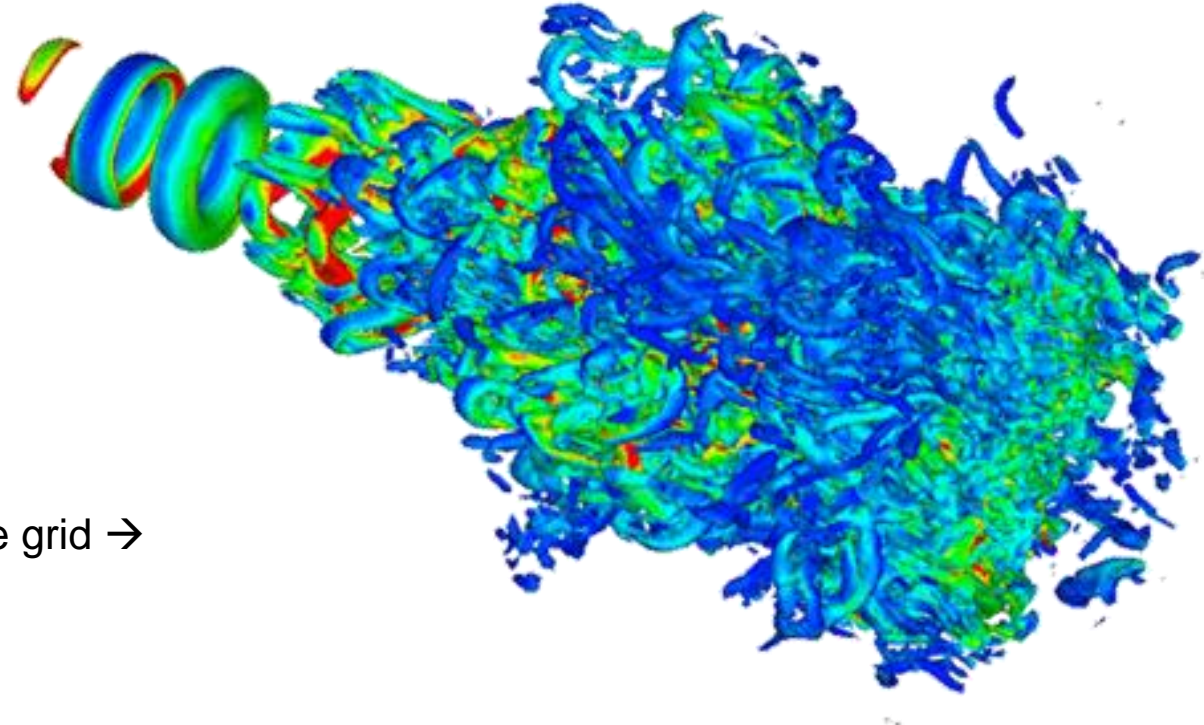
- Semi-Implicit Method for Pressure Linked Equations (SIMPLE)
 - Originally for steady state problems
 - A guess-and-correct procedure
 - SIMPLER (SIMPLE Revised) and SIMPLEC (SIMPLE Consistent)
- Pressure Implicit with Splitting of Operators (PISO)
 - Originally for unsteady compressible flows
 - Non-iterative: using one predictor and two corrector steps
- PIMPLE
 - Hybrid SIMPLE/PISO
 - PISO internal loop with SIMPLE External loops
 - Higher stability and reliability for bigger time steps
- Coupled algorithm
 - Discretized pressure and velocity equations are solved in a single matrix
 - Implicit coupling between pressure and velocity
 - Computationally more expensive
 - More stable for low quality meshes or large time steps





Models – Turbulence

- Many engineering applications are turbulent
- Turbulence
 - Reynolds number (Re)
 - Highly transient phenomenon
 - Characterized by a wide range of eddy sizes
 - Fully resolve these eddies numerically
 - Obtain a full profile of the turbulent flow field
 - Computationally very expensive
 - Hence we require a turbulence model
- Turbulence modeling
 - Important feature is averaging
 - Scales of the flow that are not resolved by the grid \rightarrow models need to be applied
- Turbulence resolving
 - Reynolds Average Navier Stokes (RANS)-based models
 - Large eddy simulations (LES)
 - Direct Numerical Simulation (DNS)





Models – Turbulence – RANS

- Any property can be written as the sum of an average and a fluctuation → **Reynolds decomposition**

$$\tilde{\varphi} = \Phi + \varphi$$

- *The average of the fluctuating component is identically zero*

- Conservation of mass

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \tilde{\mathbf{u}}) = 0 \quad \longrightarrow \quad \frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{U}) = 0$$

- Conservation of momentum (Navier-Stokes equation)

$$\frac{\partial(\rho \tilde{u}_i)}{\partial t} + \nabla \cdot (\rho \tilde{u}_i \tilde{\mathbf{u}}) = -\frac{\partial \tilde{p}}{\partial x_i} + \nabla \cdot (\mu \nabla \tilde{u}_i) + \tilde{S}_{Mi} \quad \longrightarrow \quad \frac{\partial(\rho U_i)}{\partial t} + \nabla \cdot (\rho U_i \mathbf{U}) = -\frac{\partial P}{\partial x_i} + \nabla \cdot (\mu U_i) - \left(\frac{\partial(\rho \overline{u_i u_i})}{\partial x} + \frac{\partial(\rho \overline{v v u_i})}{\partial y} + \frac{\partial(\rho \overline{w u_i})}{\partial z} \right) + S_{Mi}$$

- Conservation of passive scalars (given a scalar \tilde{e})

$$\frac{\partial(\rho \tilde{e})}{\partial t} + \nabla \cdot (\rho \tilde{e} \tilde{\mathbf{u}}) = \nabla \cdot (k \nabla \tilde{T}) + \tilde{S}_e \quad \longrightarrow \quad \frac{\partial(\rho E)}{\partial t} + \nabla \cdot (\rho E \mathbf{U}) = \nabla \cdot (k \nabla T) - \left(\frac{\partial(\rho \overline{u e})}{\partial x} + \frac{\partial(\rho \overline{v e})}{\partial y} + \frac{\partial(\rho \overline{w e})}{\partial z} \right) + S_e$$

- New unknowns

- 6 turbulent stresses ($\rho \overline{u u}$, $\rho \overline{v v}$, $\rho \overline{w w}$, $\rho \overline{u v}$, $\rho \overline{v v}$, $\rho \overline{w v}$, $\rho \overline{u w}$, $\rho \overline{v w}$, $\rho \overline{w w}$)

- 3 turbulent fluxes ($\rho \overline{u e}$, $\rho \overline{v e}$, $\rho \overline{w e}$)

- Additional equations based on empirical observations → **Turbulence Modeling** (e.g. k- ϵ , k- ω , ...)

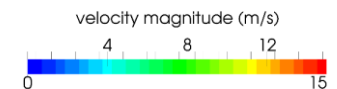
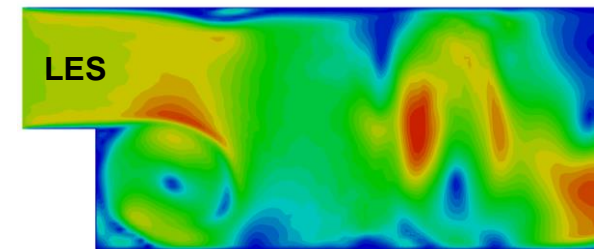
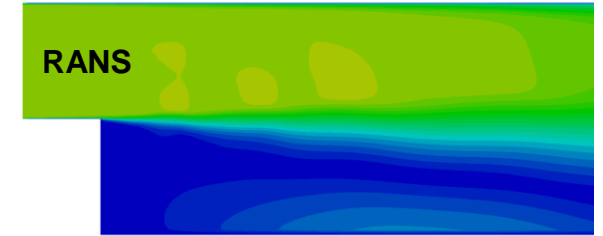
- Using PDE's for the turbulent stresses and fluxes

- Suitable for steady-state problems



Models – Turbulence – LES

- Physical representation
 - large eddies of the flow are dependent on the geometry
 - The smaller eddies are more universal
- Mathematical approach
 - Velocity field separated into a resolved and sub-grid part using a filter function
 - Convolution of a function with a filtering kernel
 - Large eddies explicitly resolved by the grid
 - Small eddies handled implicitly by sub grid-scale model (SGS)
- most practical (and commercial) implementations of LES use the grid itself
 - No explicit filtering is needed
- Sub grid-scale turbulence models usually employ the Boussinesq hypothesis
 - Turbulent stresses are related to the mean velocity gradients

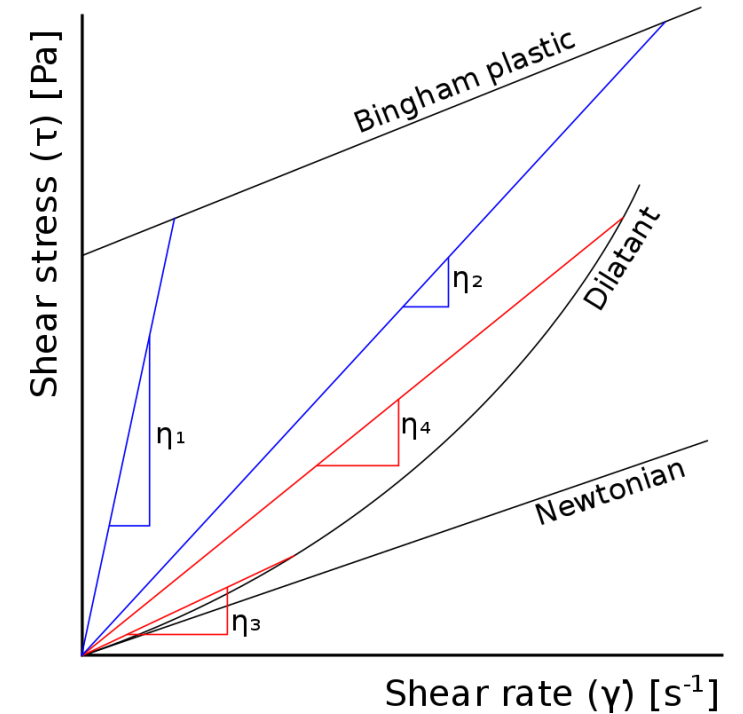


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Models – Viscosity

- Viscosity
 - Intensive property of a fluid
 - Fluid internal resistance to motion or deformation
 - Viscous fluids are less willing to flow than the less viscous fluids
- Newton's law of viscosity
 - Relationship between a fluid's shear stress and shear rate when subjected to mechanical stress
- Types of viscosity
 - Dynamic (absolute) viscosity: fluid's internal resistance to flow
 - Kinematic viscosity: ratio of dynamic viscosity to density
 - Apparent (steady shear) viscosity: shear stress ratio to shear rate
- Fluids
 - Newtonian
 - Non-Newtonian
 - Viscosity not constant (e.g. shear-rate dependent)

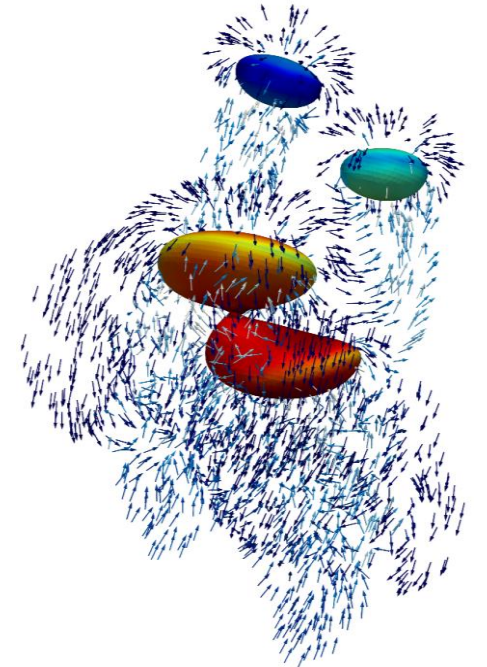


https://en.wikipedia.org/wiki/Apparent_viscosity



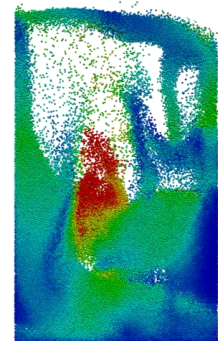
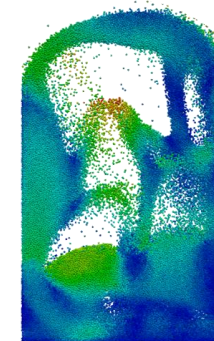
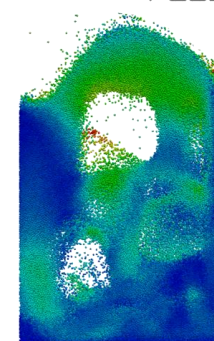
Models – Multiphase

- Simultaneous flow of materials in different phases
 - Multiple components can be present per phase.
 - gas-liquid, gas-solid, liquid-solid, liquid-liquid and three-phase flows
- Phases (visual appearance)
 - Separated: the boundary between phases is described in detail
 - Mixed: dispersed particles as well as semi-continuous interfaces exist together
 - Dispersed: one phase is dispersed in a continuous phase
- Modeling approaches
 - Lagrangian
 - Tracking individual point particles during their movement
 - More suitable for dispersed configuration
 - Studying particle flows, e.g. in Discrete Element Method (DEM)
 - Eulerian
 - Observing fluid behavior in a given control volume
 - More suitable for fluid-fluid multiphase flows
 - CFD approaches such as Euler-Euler and Volume of Fluid



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PHASIC
FLOW
PLUS

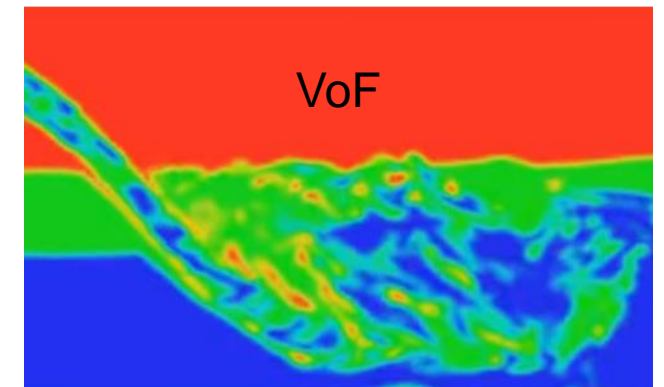
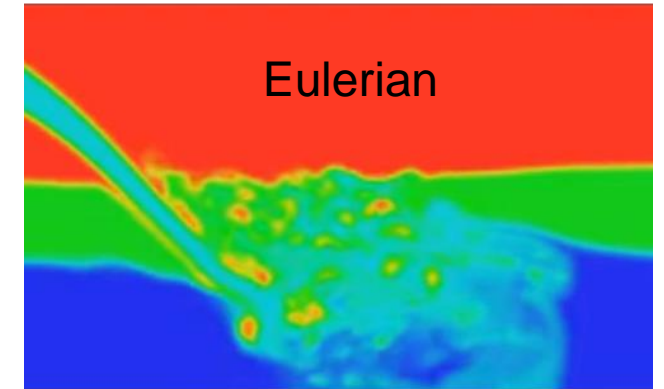


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Models – Multiphase – Eulerian

- Euler-Euler approach (Multi-fluid model)
 - All phases are treated as continuous
 - The phases interact through the drag and lift forces
 - Concept of phasic volume fractions
 - Continuous functions of space and time
 - Their sum equal to one
 - A transport equation for the volume fraction is solved
 - Individual conservation equations solved per phase
- Volume of Fluid (VoF) method
 - Fraction function (C)
 - $C=1$, the control volume is completely filled with the chosen phase
 - $C=0$, the control volume is filled with a different phase
 - $0 < C < 1$, the interface between phases is present inside the control volume
 - The flow domain is modeled on a fine grid
 - The interface to be resolved
 - To track the interface between phases:
 - All field variables are shared between the phases
 - The transport equations are solved for mixture properties
 - An advection equation for the fraction function C is solved



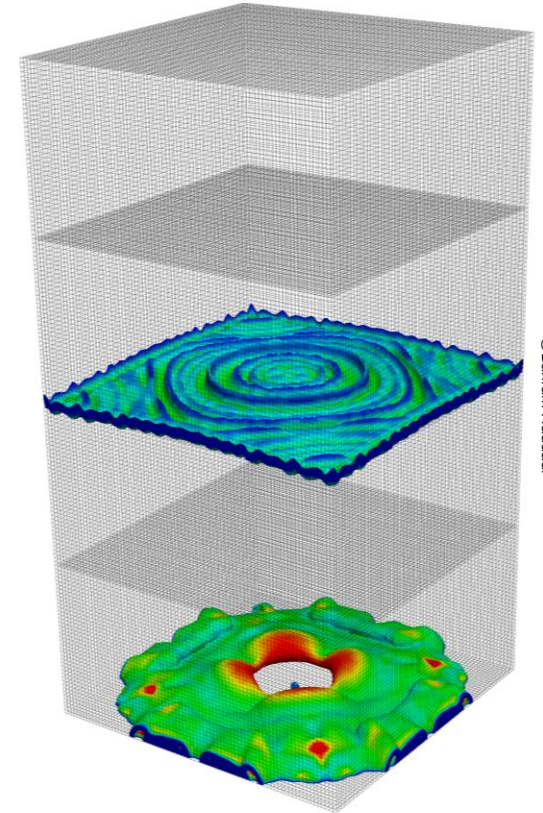
Clemens Lischka, www.youtube.com



Parallel Processing

- Simultaneous use of more than one processor
 - CPU
 - GPU
- Save time and cost
- Computational load will be distributed among processors
 - domain
 - calculations
- Load distribution
 - Shared memory: the whole system seen as a single computer
 - Distributed memory: individual computers connected through network each seen as a computational node

	Shared Memory Multiprocessor	Distributed Memory Multicomputer
Memory	Data is saved in a global memory that can be accessed by all processors	Each computer has a local memory and a processor can only access its local memory
Data transfer between processors	The sender processor simply needs to write the data in a global variable and the receiver can read it	Message is sent explicitly from one computer to another using a message passing library, e.g. Message Passing Interface (MPI)



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